Digital Control Systems MAE/ECEN 5473

Control design in frequency domain

Oklahoma State University

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What we have covered so far:

#### Conventional (classical): Analysis Design

#### Modern (state space): Analysis Design

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# Conventional control design

- Based on Z-transform
- Open-loop system vs. closed-loop system in block diagrams

Objective: design G<sub>D</sub>(z) such that the closed-loop system satisfies certain performance requirements.

# Performance specifications

- Transient performance
- Steady-state performance
- Standard testing signal for r (input):
  - (unit) step function
  - ramp function
  - acceleration function
  - sinusoidal funcation

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#### Transient performance

Given a step input (assume all initial conditions are 0):

- delay time  $(t_d)$ : time when the output reaches  $\frac{C_F}{2}$
- ▶ rise time  $(t_r)$ : time the output takes to rise from 10%  $C_F$  to 90%  $C_F$
- peak time  $(t_p)$ : time when the output reaches the first peak
- Maximum shoot: the output at  $t_p$ ,  $y(t_p)$
- Overshoot percentage:  $\frac{y(t_P) C_F}{C_F} \times 100\%$
- Settling time (t<sub>s</sub>): time required for the output to settle within 2% (5%) of the final value C<sub>F</sub>

Map damping ratio/natural frequency in Z-domain

Recall  $z = e^{Ts}$ . Since  $s = \sigma + j\omega$ ,

$$z = e^{T(\sigma + j\omega)} = e^{T\sigma} e^{jT\omega}.$$
 (1)

Consider the 2nd order system:  $G(s) = \frac{w_n^2}{s^2 + 2\zeta w_n s + w_n^2}$ 

- Poles:  $s_{1,2} = -\zeta w_n \pm \sqrt{1-\zeta^2} w_n j w_n, \zeta$ ?
- From (1),  $z_{1,2} = e^{Ts_{1,2}} = e^{-\zeta w_n T} e^{\pm T \sqrt{1-\zeta^2} w_n j}$ .
- Thus,  $|z_{1,2}| = e^{-\zeta w_n T}$ ,  $\angle z_{1,2} = \pm T \sqrt{1-\zeta^2} w_n$ .
- Given  $\zeta$ ,  $w_n$ , we can find the corresponding poles in DT:  $z_{1,2} = e^{-\zeta w_n T} e^{\pm T \sqrt{1-\zeta^2} w_n j}$ .

Given a z-domain pole  $z = r \angle \theta$  (complex poles), find the corresponding  $\zeta$ ,  $w_n$ . Also compute the time constant  $\tau = \frac{1}{\zeta w_n}$  in terms of  $T, r, \theta$ . Hint: Start with  $r = e^{-\zeta w_n T}$  and  $\theta = T \sqrt{1 - \zeta^2} w_n$  to solve for  $w_n$  and  $\zeta$  in terms of  $r, \theta, T$ .

# Steady-state (SS) performance

- Measured by SS errors (use Final value theorem (FVT), assume FVT can be applied)
- Type 0, I, II, ... systems:
  - Type 0: finite SS error for step input, infinite error for higher order inputs (ramp, acceleration, ...)
  - Type I: 0 SS error for step input, finite SS error for ramp input, infinite SS error for higher order input (acceleration, ...)

input signal	step	ramp	acceleration	higher order
Type 0	finite error	$\infty$	$\infty$	$\infty$
Type I	0	finite error	$\infty$	$\infty$
Type II	0	0	finite error	$\infty$

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#### As the system type increases

- better SS performance
- difficult to stabilize

# System type

Suppose that the open-loop PFT is 1/(z-1)<sup>N</sup> B(z)/A(z), where B(z) and A(z) contain no zero or pole at z = 1. The closed-loop system is type N, provided that the closed-loop system is stable.

• Example for N = 1, 2 with the block diagram

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### Root locus based control design

The PTF of the closed-loop system is given by



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The PTF of the closed-loop system is given by

$$\frac{C(z)}{R(z)} = \frac{KG(z)}{1 + KGH(z)}$$

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where G(z) = Z(G(s)) and GH(z) = Z(G(s)H(s)).

## Root locus based control design

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where 
$$G(z) = Z(G(s))$$
 and  
 $GH(z) = Z(G(s)H(s))$ .

The characteristic equation of the PTF is given by

$$\star : 1 + KGH(z) = 0, \quad KGH(z) = \frac{num(z)}{den(z)}.$$
 (2)

Root locus (RL) is the plot of the locus of the roots of  $\star$  in the *z*-plane as a function of *K*.

- The construction procedure of RL in DT is the same as in CT.
- The stability region in DT is the unit circle while in CT it is the open left half plane.

1. RL originates from the poles of GH(z) = 0 (den(z) = 0) and terminate at the zeros of GH(z) = 0 (num(z) = 0) (including  $\infty$ ). why?

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$$1+$$
KGH $(z)=0 \Rightarrow 1+$ K $rac{num(z)}{den(z)}=0 \Rightarrow den(z)+$ Knum $(z)=0$ 

RL starts at  $K = 0 \Rightarrow den(z) = 0 \Rightarrow$  poles of GH(z)RL terminates at  $K = \infty \Rightarrow num(z) = 0 \Rightarrow$  zeros of GH(z)

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 The RL on the real axix lies in a section of the real axis that is to the left of an odd number of poles and zeros on the real axis.

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- The RL on the real axix lies in a section of the real axis that is to the left of an odd number of poles and zeros on the real axis.
- 3. The RL is symmetric w.r.t. the real axis.

# Continued: RL drawing rules (only important ones)

- 4. The number of asymptotes of RL equals  $n_p n_z$ ,  $n_p$ ,  $n_z$  are the number of the poles and zeros of GH(z), respectively. The angles of the asymptotes are  $\frac{(2k+1)\pi}{n_p n_z}$ ,  $k = 0, 1, \cdots$ .
- 5. The asymptotes intersect the real axis at  $\sigma = \frac{\sum \text{poles of } GH(z) \sum \text{zeros of } GH(z)}{n_p n_z}$
- 6. The breakaway points are given by the roots of  $\frac{dGH(z)}{dz} = 0$ .

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Example: 
$$KGH(z) = \frac{K0.368(z+0.717)}{(z-1)(z-0.368)}$$

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# Design via RL

#### Principle

Consider a closed-loop system and its characteristic equation 1 + KD(z)G(z) = 0. Then a point  $z_a$  is on the RL if  $1 + KD(z_a)G(z_a) = 0$  for some K and D(z).

$$1 + KD(z_{a})G(z_{a}) = 0 \Leftrightarrow \begin{cases} K = \frac{1}{|D(z_{a})G(z_{a})|} \\ \angle D(z_{a})G(z_{a}) = \pm 180^{\circ} \end{cases}$$
(3)

Design D(z) such that the closed-loop poles satisfy certain requirements, such as a desired damping ratio and settling time.

Example (MATLAB code):  $G_p(s) = \frac{1}{s(s+2)}$ 

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## Control design based on frequency response

In CT, given a stable LTI system G(s) with sin(wt) as input:

$$x_{ss}(t) = \underbrace{|G(j\omega)|}_{magnitude} \sin(wt + \underbrace{\angle G(j\omega)}_{phase \ shift})$$

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Control design based on frequency response

In CT, given a stable LTI system G(s) with sin(wt) as input:



In DT, given a stable LTI system G(z) with sampled sin(wt) as input:



Recall  $z = e^{Ts}$  and let  $s = j\omega$ .

# Control design based on frequency response

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In DT, given a stable LTI system G(z) with sampled sin(wt) as input:



Recall  $z = e^{Ts}$  and let  $s = j\omega$ .

Observation

 $G(z)|_{z=e^{j\omega T}}$  provides magnitude and phase information of the frequency response of G(z).

#### w-plane

- Evaluating  $G(e^{j\omega T})$  can be complex.
- Simplify the procedure by transforming the design from z-domain to w-domain via the "bilinear" transformation

$$w = \frac{2}{T} \frac{z - 1}{z + 1} \Leftrightarrow z = \frac{1 + \frac{T}{2}w}{1 - \frac{T}{2}w}$$

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w-plane and s-plane are different. Why?

Once G(z) is transformed to  $G(w) = G(z)|_{z = \frac{1 + \frac{T}{2}w}{1 - \frac{T}{2}w}}$ , it may be

treated as a conventional TF in w-domian. Conventional frequency response based designs can be applie to G(w).

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- By replacing w = jν, we can draw the Bode plot for G(w)|<sub>w=jν</sub>.
- ▶ However, the frequency axis in the w-plane is distorted:

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$$w = j\nu = \frac{2}{T}\frac{z-1}{z+1}\Big|_{z=e^{j\omega T}} = j\frac{2}{T}\tan\frac{\omega T}{2}$$
$$\nu = \frac{2}{T}\tan\frac{\omega T}{2}$$

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$$\nu = \frac{2}{T}\tan\frac{\omega T}{2} \qquad \qquad \nu = \frac{2}{T}\tan\frac{\omega T}{2} \approx \frac{2}{T}\frac{\omega T}{2} = \omega$$

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 $\nu$  : ficticious frequency in w-plane,  $\omega$ : true frequency

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 $\nu$  : ficticious frequency in w-plane,  $\omega$ : true frequency

#### Example

If the design requirement says a bandwidth  $\omega_b$ , the corresponding bandwidth in the w-plane is  $\nu_b = \frac{2}{T} \tan \frac{\omega_b T}{2}$ .

# Phase lead/lag design (design in w-plane)

- 1. Obtain G(z) and transform it to  $G(w) = G(z)|_{z = \frac{1 + \frac{T}{2}w}{1 - \frac{T}{2}w}}}$ .
- 2. Substitute  $j\nu$  as w in G(w)and obtain the Bode plot  $|G(j\nu)|$  and  $\angle G(j\nu)$ .
- Use the conventional design methods for CT systems to determine G<sub>D</sub>(w) such that the open-loop system G<sub>D</sub>(w)G(w) satsifies design specifications (e.g., phase margin, gain margin, DC gain condition)
- 4. Transform  $G_D(w)$  to  $G_D(z) = G_D(w)|_{w=\frac{2}{T}\frac{z-1}{z+1}}$ .
- 5. Realize  $G_D(z)$  via a computational algorithm, e.g., convert it to a difference equaiton or a SS representation
- \*: Frequency axis in the w-plane is distorted. what is the relationship?

# Review of Bode plot in CT

#### Definition

- how to sketch
- Gain and phase margins

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Example

# Phase lag compensation in DT

Compensator:  $G_D(w) = k_D \frac{1 + \tau w}{1 + \beta \tau w}$ ,  $(k_D > 0, \tau > 0, \beta > 1)$ 

Pole:

Zero:

▶ Bode plot for  $G_D(w)$ 

 Magnitude and phase plots Closed-loop system diagram

Open-loop system  $G_D(w)G(w) = k_D \frac{1+\tau w}{1+\beta \tau w} G(w)$ .

#### Determine the compensator

Determine k<sub>D</sub> based on gain conditions:

- Design τ and β to yield a desired phase margin φ<sub>m</sub> (given).
   Let G<sub>1</sub>(w) = k<sub>D</sub>G(w) (known).
  - Create the bode plot of  $G_1(w)$  (magnitude and phase)
  - Determine the frequency w<sub>w1</sub> at which the phase plot of G<sub>1</sub>(jν) has a phase angle of −180° + φ<sub>m</sub> + 5° ~ 12° (to compensate for the phase lag of G<sub>D</sub>(w))
  - *w<sub>w1</sub>* is the frequency at which the phase margin occurs (after compensation).
  - Set  $\frac{1}{\tau} = 0.1 w_{w_1}$  (far away from  $w_{w_1}$ )
  - At  $w_{w_1}$ , we would like  $|G_D(jw_{w_1})G_1(jw_{w_1})| = 1$ , which leads to

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## Convert to z-domain

• Once  $G_D(w)$  is solved from the previous procedure,

$$G_D(z) = G_D(w)|_{w = \frac{2}{T} \frac{z-1}{z+1}} = k_D \frac{(1 + \frac{2\tau}{T})z + 1 - \frac{2\tau}{T}}{(1 + \frac{2\beta\tau}{T})z + 1 - \frac{2\beta\tau}{T}}.$$
 (4)

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### Phase lead compensators in DT

Compensator:  $G_D(w) = k_D \frac{1+\tau w}{1+\alpha \tau w}$ ,  $(k_D > 0, \tau > 0, \alpha > 1)$ 

Pole:

Zero:

- Bode plot for  $G_D(w)$
- Magnitude and phase plots

Open-loop TF: 
$$G_D(w)G(w) = k_D \frac{1+\tau w}{1+\alpha \tau w} G(w) = \frac{1+\tau w}{1+\alpha \tau w} \underbrace{[k_D G(w)]}_{G_1(w)}$$

# Design procedure

- Determine k<sub>D</sub> based on a given gain or static velocity error constant.
- Create bode plots of  $G_1(w)$  and evaluate the phase margin.
- Determine the necessary phase lead angle \u03c6 to be added to the system (based on the phase margin requirement)
- $\blacktriangleright$  Add 5  $\sim$  12° to  $\phi$  to compensate for the shift of the crossover frequency

• 
$$\phi + 5 \sim 12^{\circ} = \phi_m$$
. From  $\sin \phi_m = \frac{1-\alpha}{1+\alpha}$ , solve  $\alpha$ .

Select 
$$v_m = \frac{1}{\sqrt{\alpha \tau}}$$
 as the gain crossover frequency. How?

• Set 
$$\tau = \frac{1}{\sqrt{\alpha v_m}}$$
.

Check the phase margin and repeat the process by modifying pole/zero locations.

• Convert  $G_D(w)$  into z-domain using  $w = \frac{2}{T} \frac{z-1}{z+1}$ .

# Example (MATLAB code)

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